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1 Vortex-Induced-Vibration of Jack-ups with Cylindrical Legs in Multiple Modes

2 Sudheesh Ramadasan,

3 Marine Technology, Newcastle University in Singapore, 537 Clementi Road #06-01, SIT@NP Building,  
4 Singapore 599493

5 Longbin Tao\*,

6 Department of Naval Architecture, Ocean & Marine Engineering, University of Strathclyde, Glasgow G4  
7 OLZ, United Kingdom

8 Arun Kr Dev,

9 Marine Technology, Newcastle University in Singapore, 537 Clementi Road #06-01, SIT@NP Building,  
10 Singapore 599493

11 **Abstract**

12 A simple mathematical model was developed based on the single-degree-of-freedom analogy and  
13 principle of conservation of energy evaluating various modes of Vortex-Induced-Vibration (VIV) of a  
14 jack-up with cylindrical legs in steady flow. Mass ratio, damping ratio and mode factor were found to be  
15 the important parameters controlling the inline and cross flow VIV and radius of gyration for the yaw  
16 VIV. Criteria for the initiation of the three VIV modes were developed for the cases of a single 2D  
17 cylinder, four rigidly coupled 2D cylinders in rectangular configuration and a jack-up experiencing

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\* Corresponding author. Tel: +44 (0)141 548 3315; Email: [longbin.tao@strath.ac.uk](mailto:longbin.tao@strath.ac.uk)

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1 uniform flow. The model tests demonstrated that the jack-up with cylindrical legs experienced cross flow  
2 and yaw VIV in uniform flows, with amplitude ratios greater than  $0.1D$ . Further, there was considerable  
3 overlap of the lock-in ranges and coupling at higher current speeds of the aforementioned modes making  
4 the jack-up practically redundant throughout the operating currents. The analysis of the mean inline  
5 responses of the model revealed drag amplification due to the VIV. The test results validated the developed  
6 VIV model, VIV criteria and the importance of mass ratio in suppressing VIV. The mathematical method  
7 will enable practising engineers to consider the effect of VIV in jack-up designs.

8 Keywords, Jack-up, Vortex-induced-vibration (VIV), VIV criteria, VIV suppression

## 9 1. Introduction

10 Flow around circular cylinders is one of the extensively researched subjects in fluid mechanics. Alternate  
11 shedding of vortices about the cylinders due to flow separation can cause Vortex-Induced-Vibration  
12 (VIV), which can eventually lead to structural yield and fatigue failures. Comprehensive reviews on VIV  
13 can be found in Sarpkaya and Isaacson (1981), Blevins (2001), Sarpkaya (2004), Williamson and  
14 Govardhan (2004) and Sumer and Fredsøe (2006).

15 VIV in water is characterised by low mass ratio, added mass and significant fluid damping. King et al.  
16 (1973) investigated the nature of VIV in water by means of an exhaustive test programme using model  
17 piles and observed VIV along both inline and cross flow directions. Khalak and Williamson (1997a)  
18 conducted experiments with an elastically mounted rigid cylinder in water and found an upper branch of  
19 VIV with large vibration amplitudes owing to very low mass ratio and damping. It was observed that  
20 oscillation frequency kept increasing above the natural frequency throughout the excitation regime and  
21 considerable lift and drag amplification during lock-in vibrations. Khalak and Williamson (1997b)  
22 presented the results of tests performed to separately determine the effect of mass ratio and damping ratio.

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1 It was reported that the overall range of excitation as well as the shape of the response curve is determined  
2 by the mass ratio while the level of response is characterised by the mass damping parameter. Govardhan  
3 and Williamson (2004) found that the vibration frequency during lock-in is primarily dependent on mass  
4 ratio and vibration frequency can reach remarkably large values when mass ratio is in the order of unity.  
5 The authors proposed an expression for the lower branch vibration frequency and confirmed the existence  
6 of a critical mass ratio below which the structure will vibrate till infinite flow speed in the upper branch.  
7 This revealed that the added mass of cylinders during large amplitude cross flow vibration is negative.  
8 Blevins and Coughran (2009) conducted tests on VIV of an elastically supported cylinder in water and  
9 observed that the lock-in range increased with a decrease in mass ratio and for light cylinders, the vibration  
10 frequency increased up to 50% above cylinder's natural frequency. Vandiver (2012) proposed an  
11 alternative damping parameter ( $C^*$ ) to overcome the limitation of mass damping parameter and argued  
12 that the alternative might be used to characterise VIV at all reduced velocities in the lock-in range and  
13 multiplied with the amplitude ratio to calculate the lift coefficient. It was also noted that inline cylinder  
14 vibrates at twice the shedding frequency.

15 Multi-cylinder structures are extensively used in marine and offshore industries, typical concepts being  
16 jacket platforms, semi-submersibles, riser bundles and jack-up platforms. Figure 1 displays a typical jack-  
17 up with cylindrical legs performing soil investigation. Floating multi-cylinder structures like semi-  
18 submersibles and TLPs are found to experience Vortex Induced Motion (VIM), a low frequency  
19 equivalent of VIV. The mass ratio of unity and low slenderness ratio of the cylinders can be regarded as  
20 the main difference of VIM with jack-up undergoing VIV. However, as previous studies are limited for  
21 the VIV of a jack-up, the available literature on VIM and VIV of rigidly coupled multiple cylinders can  
22 be used effectively to draw relevant insights on the former. A comprehensive review of the research  
23 developments on VIM was presented by Fujarra et al. (2012). Gonçalves et al. (2011b) conducted  
24 experimental studies on VIM with a scaled model of the semi-submersible and found that inline, cross

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1 flow and yaw motions were experienced. It was also observed lock-in for both the cross flow and yaw  
2 motions and the later was named as vortex-induced-yaw motion (VIY). It was further stated that VIY  
3 occurred when vortex shedding frequency about the columns approached the natural frequency of yaw of  
4 the semi-submersible. Liang et al. (2017) carried out a comprehensive numerical study complimented by  
5 experimental measurements on the VIM of a deep draft semi-submersibles to examine the characteristics  
6 of vortex shedding processes and their interactions due to multiple cylindrical columns. Gonçalves et al.  
7 (2018) conducted experiments on deep draft semi-submersibles models to investigate the effects of the  
8 column shape and the surface roughness. It was found that the circular column experienced higher VIM  
9 amplitudes in both transverse and yaw motions and confirmed the existence of lock-in ranges for both  
10 modes. It was also observed dynamic amplification of drag, lift and yaw moment coefficients during lock-  
11 in. Gonçalves et al. (2011a) conducted VIV experiments with cantilevered bar and pivoted pendulum of  
12 low mass and aspect ratios, and compared the results with the published literature on VIM to establish  
13 similarity between VIM and VIV. It was observed similarity in trends and values between both the  
14 phenomena and that higher aspect ratios experienced higher amplitudes due to enhanced vortex  
15 correlation.

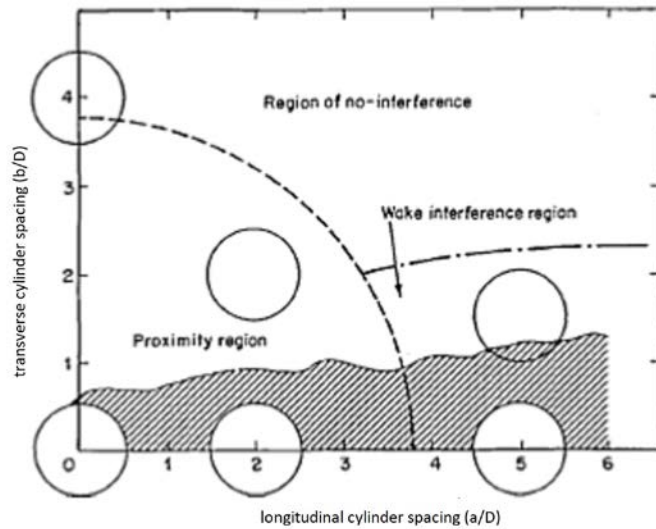


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1        Figure 1. Jack-up with cylindrical legs (Ms Cybermarine Technologies Pte. Ltd.)

2        The spacing between the cylinders is a significant parameter for multi-cylinder VIV as the cylinders are  
3        strongly influenced by wake and proximity interference effects. The flow around the downstream  
4        cylinders is highly influenced by the wake of the upstream cylinders. Sumner (2010) has presented an  
5        exhaustive review of the literature on the flow around two identical circular cylinders in a steady flow,  
6        covering the three main configurations namely tandem, side by side and staggered. Bearman (2011) has  
7        reviewed the recent research on the VIV of isolated circular cylinders and circular cylinders in tandem  
8        arrangement. Zdravkovich (1985) conducted extensive wind tunnel experiments with two identical  
9        cylinders in tandem, side by side and staggered arrangements covering the proximity interference, wake  
10       interference and no interference regions as depicted in Figure 2. Based on the experimental study,  
11       Zdravkovich (1985) concluded that the coupling between cylinders disappeared when the transverse pitch  
12       ratio was above 4, and the wake interference gradually diminished when the longitudinal pitch ratio was  
13       greater than 7. Wang et al. (2013) conducted an experimental study of flow around four circular cylinders  
14       in a square configuration and reconfirmed that depending on the pitch ratio, the inline flow can be broadly  
15       classified as shielding regime, shear layer reattachment regime and vortex impinging regime. It was  
16       observed that for transverse spacing ratios above 4, the four cylinder array could be regarded as two  
17       isolated parallel rows of two cylinders in tandem. It was further revealed that for large pitch ratios ( $P/D >$   
18       5), the vortex shedding from all four cylinders is fully synchronised with constant frequency and definite  
19       phase relationships. Assi et al. (2010) studied the wake induced vibration (WIV) response of a downstream  
20       cylinder for various longitudinal pitch ratios and found that the downstream cylinder experienced  
21       vibrations with increasing amplitudes at higher reduced velocities for the pitch ratios less than 8. For pitch  
22       ratios greater than 8, however, the WIV was seen progressively reduced, and the response amplitude peak  
23       corresponded to that of VIV resonance.



1

2 Figure 2. Cylinder interference regions (Zdravkovich, 1985)

3 Jiang (2012) numerically studied flow induced transverse vibrations of two tandem cylinders between two  
 4 parallel walls and effect of pitch ratios ranging from 1.1 to 10. It was found that the cylinders decouple  
 5 and behave as two isolated cylinders for larger pitch ratios above 8. Han et al. (2015) numerically  
 6 investigated the flow induced vibration of four uncoupled identical circular cylinders in a square  
 7 arrangement subjected to uniform flow. Dual resonance or cylinder synchronised vibrations were  
 8 identified along both inline and cross flow directions during lock-in for a pitch ratio of 5. Zhao and Cheng  
 9 (2012) performed numerical simulation of VIV of four rigidly coupled square cylinders in a square  
 10 configuration with a pitch ratio of 3. It was reported that there were two modes of vortex shedding,  
 11 symmetrical and synchronised. In the symmetrical mode, cylinders on either side were found to shed  
 12 vortices symmetrically about the longitudinal centre line. The synchronised mode was found to cause  
 13 aggressive lock-in vibrations due to the synchronisation of vortex shedding about all the four cylinders.

14 Self-elevating platforms or jack-ups with cylindrical legs are usually deployed in shallow and intermediate  
 15 waters. Cylindrical legs, typically having diameters from 0.50m to 4.00 m can potentially lead to large  
 16 resonant vibration and lock-in in lateral and yaw directions of the jack-ups when the vortex shedding

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1 frequency approaches the unit's natural frequency. Such VIV can amplify the mean drag acting on the  
2 legs, resulting in high static and cyclic stresses and eventually lead to yield and fatigue damages. Nicholls-  
3 Lee et al. (2013) pointed out the instability of the jack-ups in high tidal currents and associated VIV  
4 causing large topside motions leading to abortion of operations. Thake (2005) described the VIV  
5 experienced by jack-ups working in deep and fast currents and further stated the lack of established design  
6 guidelines to address this issue and the need for developing new methods. The potential modes of VIV of  
7 the jack-up are illustrated in Figure 3.

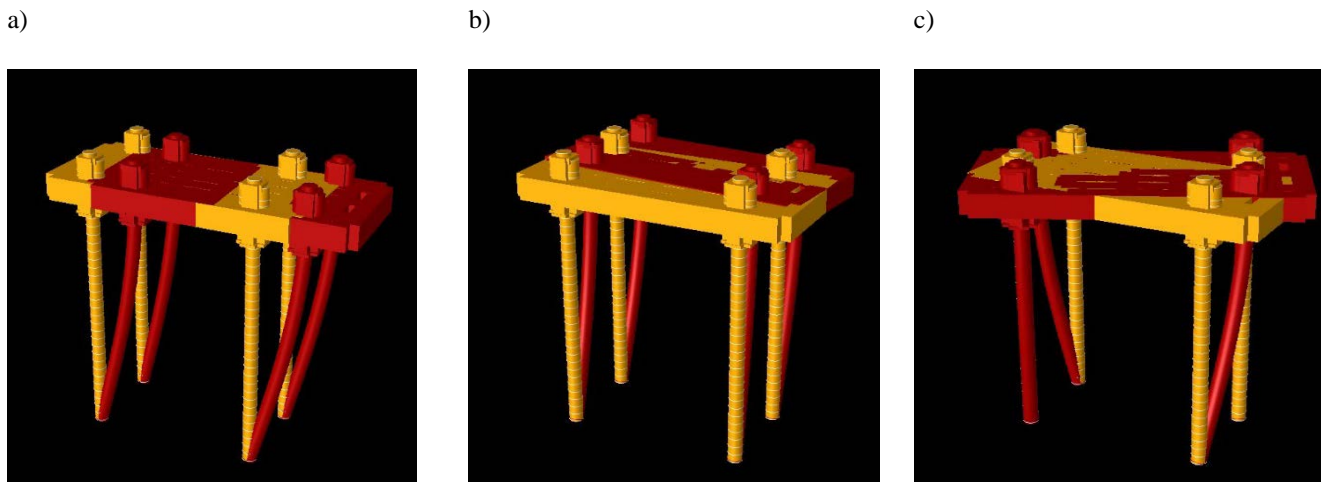


Figure 3. Jack-up VIV modes, a) in-line (surge), b) cross flow (sway), c) yaw (torsional)

8 The harmonic model was used to develop criteria (Barltrop and Adams, 1991; Blevins, 2001) for the  
9 occurrence of VIV. It was found that various modes of VIV can be suppressed by increasing reduced  
10 damping or mass damping parameter above certain threshold values. King et al. (1973) conducted  
11 experiments on model piles in water and developed stability criteria for the occurrence of VIV; stability  
12 parameters 1.2 and 17 for in-line and cross flow mode respectively. Sakai et al. (2002) conducted  
13 experimental study on cantilever cylinders with various reduced damping and confirmed the criteria of  
14 reduced damping of 1.2 in the in-line direction. Vickery and Watkins (1964) derived the dependence of  
15 response amplitude on reduced damping by considering the energy balance between excitation and

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1 damping at resonance, and presented conditions of similarity between model and prototype for VIV  
2 experiments.

3 There are limited model test results on jack-ups available in the published literature, and almost none of  
4 them pertains to jack-up VIV. Bennett Jr and Patel (1989) tested the scaled down model of a jack-up with  
5 truss legs to understand the dynamics in regular waves and confirmed that the behaviour was similar to a  
6 classical SDOF (single-degree-of-freedom) system. Johnson and Patel (1992) and Grundlehner (1997)  
7 presented the tests carried out with the free and restrained jack-up model in the elevated condition to  
8 measure and verify the dynamic response. The results of the fixed and free tests were compared, and the  
9 dynamic amplification factor was obtained. Cammaert et al. (2014) carried out model tests of an arctic  
10 jack-up with cylindrical legs to verify the ice loads and leg sheltering factor. Journee et al. (1988) had  
11 carried out experiments with simplified jack-up models with cylindrical legs and investigated the  
12 hydrodynamic and structural nonlinearities in the fluid-structure interaction.

13 The main objectives of the paper are to understand the VIV of the jack-up with multi-cylindrical legs by  
14 means of model tests and to develop simple mathematical models to evaluate the various VIV modes. A  
15 set of criteria are then proposed for predicting the occurrence of the various VIV modes of jack-ups. The  
16 significance of such a simplified mathematical approach is to enable practising engineers to account for  
17 the effect of the VIV in the design stage of jack-ups. The comprehensive experimental results presented  
18 in this paper will also serve as a benchmark for the future research of jack-up VIV.



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## 1 2. Theoretical Evaluation

### 2 2.1. 2D Circular Cylinders VIV in Steady Flow

#### 3 2.1.1. Inline/ Surge VIV of a Single Cylinder

4 The inline resonant response amplitude,  $x_o$  of a lightly damped linear mass spring SDOF cylinder under  
5 dynamic excitations can be expressed as (Ramadasan et al., 2018),

$$6 \quad x_o = \frac{F_d}{C\omega_N} \quad (1)$$

7 where  $F_d$ ,  $C$  and  $\omega_N$  represent the amplitude of the excitation force, damping coefficient and natural  
8 angular frequency respectively. The product  $C\omega_N$  is the dynamic damping stiffness and controls the  
9 resonance amplitude. The oscillatory drag excitation ( $F_d$ ) can be expressed as,

$$10 \quad F_d = \frac{1}{2}\rho C_d D U^2 L \quad (2)$$

11 where  $\rho$ ,  $C_d$ ,  $D$ ,  $U$  and  $L$  denote the density of the fluid, oscillatory drag coefficient, diameter of the  
12 cylinder, flow velocity and length of the cylinder respectively.

13 The flow velocity can be expressed in terms of the Strouhal relationship as,

$$14 \quad U = \frac{f_v D}{St} = \frac{\omega_v D}{2\pi St} \quad (3)$$

15 where  $f_v$ ,  $\omega_v$  and  $St$  represent the vortex shedding frequency, vortex shedding angular frequency and  
16 Strouhal number respectively.

17 The damping coefficient can be expressed in terms of a damping ratio ( $\zeta$ ) as,

$$18 \quad \zeta = \frac{C}{2\sqrt{MK}} \quad (4)$$

---

1 where  $M$  and  $K$  represent the mass and inline stiffness respectively of the cylinder.

2 During inline resonance or lock-in,

3 
$$\omega_V = \frac{\omega_N}{2} = \frac{1}{2} \sqrt{\frac{K}{M}} \quad (5)$$

4 Defining mass ratio ( $m^*$ ) of a cylinder as the ratio of mass over displaced mass,

5 
$$m^* = \frac{M}{\rho \frac{\pi}{4} D^2 L} \quad (6)$$

6 Substituting Equations (2) to (6) in (1), the amplitude ratio of the inline VIV response can be derived as,

7 
$$\frac{x_O}{D} = \frac{C_d}{16\pi^3 St^2 \zeta m^*} \quad (7)$$

8 It can be seen from Equation (7) that the inline response amplitude ratio of a cylinder undergoing inline  
9 VIV in a steady flow is inversely proportional to the product of mass and damping ratios. The Reynolds  
10 number (Re) dependence of the inline VIV is also visible from the presence of Strouhal number and drag  
11 coefficient in the above expression.

12 A criterion for the occurrence of the inline VIV can be derived by considering 1% amplitude ratio  
13 (Bartrop and Adams, 1991).

14 
$$\zeta m^* \leq \frac{25 C_d}{4\pi^3 St^2} \quad (8)$$

15 Considering a Strouhal number of 0.20 and a maximum stationary cylinder oscillatory drag coefficient of  
16 0.10 (Sumer and Fredsøe, 2006) for the practical Re range, the criterion for the inline VIV can be derived  
17 as

18 
$$\zeta m^* \leq 0.50 \quad (9)$$

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1 2.1.2. *Cross flow/Sway VIV of a Single Cylinder*

2 During cross flow resonance or lock-in,

3 
$$\omega_V = \omega_N = \sqrt{\frac{K}{M}} \quad (10)$$

4 Like inline response, the amplitude ratio of the cross flow VIV response ( $y_0$ ) of a 2D cylinder in a steady  
5 flow can be derived as.

6 
$$\frac{y_0}{D} = \frac{C_L}{4\pi^3 St^2 \zeta m^*} \quad (11)$$

7 where  $C_L$  represents the lift coefficient of the cylinder. The equation also shows the inverse proportionality  
8 to the product of mass and damping ratios and the Reynolds number dependence of the cross flow VIV.

9 Similar to inline VIV, the criterion for the occurrence of the cross flow VIV can also be derived by  
10 considering a 1% amplitude ratio (Barltrop and Adams, 1991).

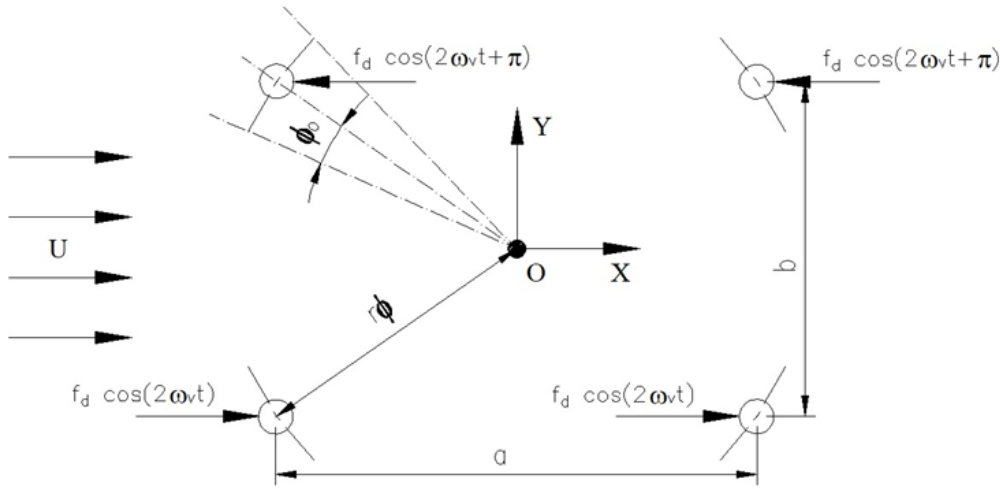
11 
$$\zeta m^* \leq \frac{25C_L}{\pi^3 St^2} \quad (12)$$

12 Considering a Strouhal number of 0.20 and a maximum stationary cylinder lift coefficient of 0.85 for the  
13 practical Re range, the cross flow VIV criterion can be derived as,

14 
$$\zeta m^* \leq 17.20 \quad (13)$$

15

1 *2.1.3. Yaw VIV of a system of four 2D cylinders in rectangular configuration under inline excitation*



2

3 Figure 4. Yaw due to inline excitation

4 Like the single cylinder VIV, the amplitude of yaw VIV response ( $\phi_O$ ) of a system of four 2D cylinders  
 5 in a steady flow under inline or drag excitation can be expressed as,

6 
$$\phi_O = \frac{M_{\phi O}}{C_{\phi} \omega_N} \tag{14}$$

7 where  $M_{\phi O}$  and  $C_{\phi}$  represent the amplitude of yaw excitation moment and yaw damping coefficient  
 8 respectively.

9 The yaw excitation due to oscillatory drag, considering the vortex synchronisation between cylinders as  
 10 shown in Figure 4 can be,

11 
$$M_{\phi O} = 2 \times \frac{1}{2} \rho C_d D U^2 L \times b \tag{15}$$

12 where  $b$  represents the cross flow or transverse cylinder spacing.

13 During resonance/lock-in under inline excitation,

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$$\omega_V = \frac{\omega_N}{2} = \frac{1}{2} \sqrt{\frac{K_\phi}{I}} \quad (16)$$

where  $K_\phi$  and  $I$  represent the yaw stiffness and yaw moment of inertia respectively.

Considering the Strouhal relationship, mass ratio, damping ratio and defining the yaw radius of gyration ( $r_\phi$ ), the amplitude ratio of yaw VIV resonant response can be derived as,

$$\frac{\phi_O}{D} = \frac{C_d b}{32 \pi^3 St^2 \zeta m^* r_\phi^2} \quad (17)$$

Equation (17) shows the inverse proportionality of the yaw response to the product of mass and damping ratios and the Reynolds number dependence of the yaw VIV. The proportionality with the leg transverse spacing and inverse proportionality with the square of yaw radius of gyration are also captured.

Similar to inline and cross flow VIV, considering a 1 % amplitude ratio to represent VIV occurrence,

$$\frac{r_\phi \phi_O}{D} \leq \frac{1}{100} \quad (18)$$

The criterion for the occurrence of yaw VIV under drag excitation can be derived as,

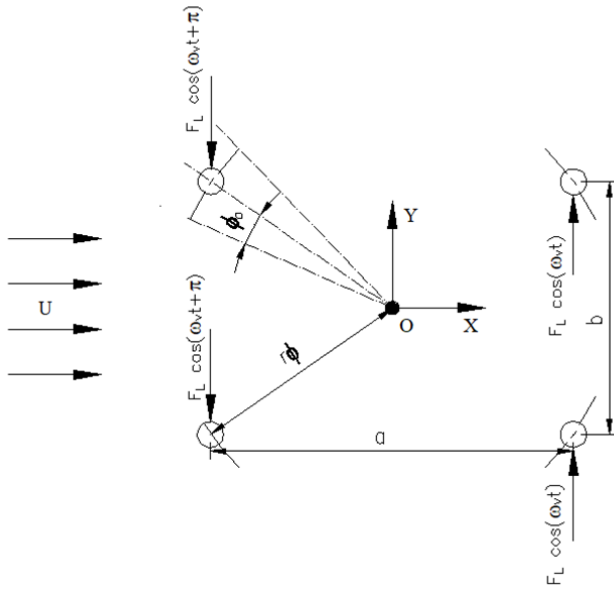
$$\zeta m^* \leq \frac{25 C_d b}{8 \pi^3 St^2 r_\phi} \quad (19)$$

Considering a Strouhal number of 0.20 and a maximum stationary cylinder oscillatory drag coefficient of 0.10 for the practical Re range, the criterion becomes,

$$\zeta m^* r_\phi \leq 0.25 b \quad (20)$$

16

1 2.1.4. Yaw VIV of a system of four 2D cylinders in rectangular configuration under cross flow excitation.



2

3 Figure 5. Yaw due to cross flow excitation

4 The yaw excitation due to lift, considering the vortex synchronisation between cylinders as shown in  
 5 Figure 5 can be expressed as,

6 
$$M_{\phi O} = 2 \times \frac{1}{2} \rho C_L D U^2 L \times a \tag{21}$$

7 where  $a$  is the inline or longitudinal cylinder spacing.

8 During resonance/lock-in under lift excitation,

9 
$$\omega_V = \omega_N = \sqrt{\frac{K_\phi}{I}} \tag{22}$$

10 Similar to yaw due to drag excitation, the amplitude ratio of the yaw VIV resonant response due to lift  
 11 excitation can be derived as,

12 
$$\frac{\phi_O}{D} = \frac{C_L a}{8 \pi^3 St^2 \zeta m^* r_\phi^2} \tag{23}$$

---

1 Equation (23) presents the inverse proportionality of the response with the product of cylinder mass ratio,  
2 damping ratios and the square of the yaw radius of gyration. Re dependence and linear proportionality  
3 with the longitudinal leg spacing are also reflected.

4 Similarly, the criterion for the occurrence of yaw VIV under lift excitation can be derived as,

5 
$$\zeta m^* \leq \frac{25 C_L a}{2 \pi^3 St^2 r_\phi} \quad (24)$$

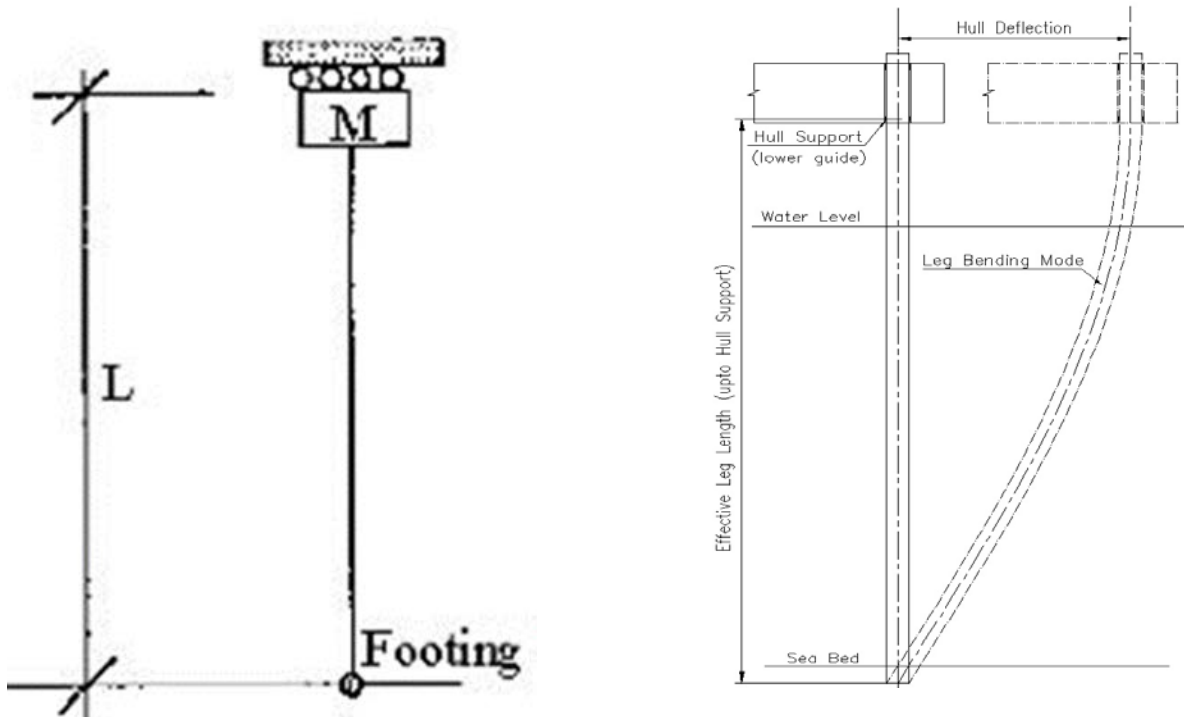
6 Considering a Strouhal number of 0.20 and a maximum stationary cylinder lift coefficient of 0.85 for the  
7 practical Re range, the criterion becomes,

8 
$$\zeta m^* r_\phi \leq 8.60 a \quad (25)$$

9

1 2.2. VIV of Jack-up with Cylindrical Legs in Steady Flow

2 Figure 6 illustrates the SDOF idealisation of jack-up and the typical leg mode shape. The jack-up is  
 3 idealised as an elevated concentrated mass in way of the hull leg interface with the legs providing flexural  
 4 stiffness against lateral deflection.



5 a) b)

6 Figure 6. Jack-up structural idealisation, a) SDOF (PANEL OC-7, 2008), b) leg mode shape

7 The mode shape or deflected profile of the leg of an independent leg type jack-up unit undergoing resonant  
 8 vibration can be idealized as a generalised sinusoidal wave. For example, for inline vibration,

9 
$$x_o(z) = \left[ \frac{X_L}{A \sin(k_L L + B) + E} \right] [A \sin(k_L z + B) + E] \quad (26)$$

10 where  $X_L$ ,  $k_L$  represent the inline response in way of the hull interface and mode shape of the leg  
 11 respectively,  $A$ ,  $B$  and  $E$  are constants depending on boundary conditions and  $z$  represents the elevation



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1 with respect to leg bottom with positive direction pointing upwards; at leg bottom,  $z = 0$  and at hull  
2 interface,  $z = L$ .

3 The effective mass per leg ( $M_{eL}$ ) of the equivalent single-degree-of-freedom (SDOF) system of a Jack-up  
4 idealised at the hull leg interface level can be expressed based on the energy principle (Barltrop and  
5 Adams, 1991) as,

$$6 \quad M_{eL} = \int_0^L m(z) \left[ \frac{A \sin(k_L z + B) + E}{A \sin(k_L L + B) + E} \right]^2 dz \quad (27)$$

7 where  $m$  represents the mass distribution along the leg.

8 As evident from the Equation (27), the contributions from the individual mass components, hull, leg,  
9 entrapped mass, added mass depend on the leg mode shape and their respective locations along the leg.

### 10 2.2.1. Inline and Cross flow VIV

11 The effective excitation force per leg ( $F_{eL}$ ) of the SDOF system of the jack-up can be expressed based on  
12 energy principle (Barltrop and Adams, 1991) as,

$$13 \quad F_{eL} = \int_0^L f_0(z) \frac{A \sin(k_L z + B) + E}{A \sin(k_L L + B) + E} dz \quad (28)$$

14 where  $f_0$  represents the distribution of the oscillatory excitation force along the leg.

#### 15 2.2.1.1. Inline / Surge VIV in Uniform Current

16 The resonant inline response ( $X_L$ ) of the SDOF can be expressed similar to Equation (1) and considering  
17 the similarity, parallelism and vortex synchronisation of the legs,

$$18 \quad X_L = \frac{F_{eL}}{c_{eL} \omega_N} \quad (29)$$

---

1 where  $C_{eL}$  represents the effective damping coefficient per leg.

2 The simplest inline VIV model for a jack-up can be developed by considering the oscillatory drag  
3 excitation as,

$$4 \quad f_{Ox} = \frac{1}{2} \rho C_d D U^2 \quad (30)$$

5 where  $f_{Ox}$  represents the distribution of the oscillatory drag force amplitude along the leg.

6 Defining the oscillatory drag force as  $F_d = f_{Ox} d'$ , Equation (29) can be simplified as,

$$7 \quad X_L = \frac{F_d}{C_{eL} \omega_N} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (31)$$

8 where  $X_L$  and  $d'$  represent the inline response of the leg in way of hull interface and the effective water  
9 depth considering leg penetration in the soil respectively.

10 Considering the Strouhal relationship, resonance/lock-in under inline excitation,

$$11 \quad \frac{X_L}{D} = \frac{C_d}{16\pi^3 St^2 \zeta m^*} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (32)$$

12 Equation (32) shows clearly the inverse proportionality with the mass damping parameter (product of  
13 mass ratio and damping ratio), the effect of leg mode shape and the Re dependence of the inline VIV.

14 Considering the 1 % amplitude ratio, the criterion for the occurrence of inline VIV of a jack-up can be  
15 derived as,

$$16 \quad \zeta m^* \leq \frac{25 C_d}{4 \pi^3 St^2} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (33)$$

17 Considering a Strouhal number of 0.20 and a maximum stationary cylinder oscillatory drag coefficient of  
18 0.10 for the practical Re range, the criterion becomes

---


$$\zeta m^* \left( \frac{d' [A \sin(k_L L + B) + E]}{\int_0^{d'} [A \sin(k_L z + B) + E] dz} \right) \leq 0.50 \quad (34)$$

The expression inside parenthesis can be defined as the mode factor (MF). The product of MF with mass damping parameter can be the effective (or modified) mass damping parameter for a jack-up. Thus the criterion becomes,

$$\zeta m^* MF \leq 0.50 \quad (35)$$

### 2.2.1.2. Cross flow / Sway VIV in Uniform Current

Like inline response, the cross flow response can also be derived considering Strouhal relationship and lock-in conditions as,

$$\frac{Y_L}{D} = \frac{C_L}{4\pi^3 St^2 \zeta m^*} \left( \frac{1}{d' [A \sin(k_L L + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (36)$$

where  $Y_L$  represents the cross flow response of the leg in way of hull interface.

It can be observed that the inverse proportionality with mass damping parameter, the effect of leg mode shape and the Re dependence are applicable also for cross flow VIV.

Considering a 1 % amplitude ratio, the criterion for the occurrence of cross flow jack-up VIV is,

$$\zeta m^* \leq \frac{25 C_L}{\pi^3 St^2} \left( \frac{1}{d' [A \sin(k_L L + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (37)$$

Considering a Strouhal number of 0.20 and a maximum stationary cylinder lift coefficient of 0.85 for the practical Re range, the criterion becomes,

$$\zeta m^* MF \leq 17.20 \quad (38)$$

18

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1 2.2.2. *Yaw VIV*

2 The total effective yaw excitation moment per leg ( $M_{e\phi_L}$ ) of the equivalent SDOF system of a Jack-up is,

3 
$$M_{e\phi_L} = \int_0^{d'} m_\phi(z) \left[ \frac{A \sin(k_L z + B) + E}{A \sin(k_L L + B) + E} \right] dz \quad (39)$$

4 where  $m_\phi$  represents the distribution of the yaw excitation moment along the leg.

5 The total yaw excitation on the jack-up ( $M_{e\phi}$ ) can be expressed as,

6 
$$M_{e\phi} = 4 M_{e\phi_L} \quad (40)$$

7 The yaw resonant response of the SDOF ( $\phi_L$ ) can be expressed similar to Equation (14) and considering  
8 the similarity and vortex synchronisation of the legs.

9 
$$\phi_L = \frac{M_{e\phi}}{C_{e\phi} \omega_N} \quad (41)$$

10 where  $C_{e\phi}$  represents the effective yaw damping coefficient of the jack-up SDOF.

11 2.2.2.1. *Yaw VIV due to inline excitation in Uniform Current*

12 The yaw excitation per unit leg length can be expressed as,

13 
$$m_\phi(z) = f_{ox}(z) \frac{b}{2} \quad (42)$$

14 where  $b$  represents the cross flow or transverse spacing of legs.

15 Considering Strouhal relationship and lock-in condition, Equation (42) can be further simplified as,

16 
$$\frac{\phi_L}{D} = \frac{C_d b}{32\pi^3 St^2 \zeta m^* r_\phi^2} \left( \frac{1}{d' [A \sin(k_L L + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] \quad (43)$$

1 Equation (43) shows the inverse proportionality of the yaw response with the product of cylinder mass  
 2 ratio, damping ratio and the square of the yaw radius of gyration, the effect of mode shape, Re dependence  
 3 and linear proportionality with the transverse leg spacing.

4 Considering a 1 % amplitude ratio, the criterion for the occurrence of yaw VIV of a Jack-up under drag  
 5 excitation can be derived as,

$$6 \quad \zeta m^* r_\phi^2 \leq \frac{25 C_d b \sqrt{a^2 + b^2}}{16 \pi^3 S t^2} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (44)$$

7 Considering a Strouhal number of 0.20 and a maximum stationary cylinder oscillatory drag coefficient of  
 8 0.10 for the practical Re range, the criterion becomes,

$$9 \quad \zeta m^* M F r_\phi^2 \leq 0.13 b \sqrt{a^2 + b^2} \quad (45)$$

10 The expression on the left-hand side (LHS) of Equation (45) is the product of the effective mass damping  
 11 parameter and the square of the yaw radius of gyration which can be termed as the effective inertia  
 12 parameter.

### 13 2.2.2.2. Yaw VIV due to cross flow excitation in Uniform Current

14 The yaw excitation per leg is,

$$15 \quad m_\phi(z) = f_{Oy}(z) \frac{a}{2} \quad (46)$$

16 where  $f_{Oy}$  and  $a$  represent the distribution of the oscillatory lift force amplitude along the leg and cross  
 17 flow or transverse spacing of legs respectively.

18 Like yaw due to drag excitation, the amplitude ratio of the yaw VIV resonant response due to lift excitation  
 19 can be derived as,

---


$$\frac{\phi_L}{D} = \frac{C_L a}{8 \pi^3 St^2 \zeta m^* r_\phi^2} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (47)$$

Equation (47) shows the inverse proportionality of the yaw response with the product of cylinder mass ratio, damping ratio and the square of the yaw radius of gyration, the effect of mode shape, Re dependence and linear proportionality with the longitudinal leg spacing.

Considering a 1 % amplitude ratio to represent VIV suppression, the criterion for Yaw VIV of a Jack-up under lift excitation can be expressed as,

$$\zeta m^* r_\phi^2 \leq \frac{25 C_L a \sqrt{a^2 + b^2}}{4 \pi^3 St^2} \left( \frac{1}{d' [A \sin(k_L z + B) + E]} \right) \int_0^{d'} [A \sin(k_L z + B) + E] dz \quad (48)$$

Considering a Strouhal number of 0.20 and a maximum stationary cylinder lift coefficient of 0.85 for the practical Re range, the criterion becomes,

$$\zeta m^* MF r_\phi^2 \leq 4.30 a \sqrt{a^2 + b^2} \quad (49)$$

11

---

1 3. Physical Experiments

2 3.1. Jack-up Model

3 A jack-up model with a scale of 1: 28 was constructed to suit the tank dimensions. The model material  
4 was selected as PVC as per Cauchy scaling to satisfy the law of similitude for structural deflections. The  
5 leg footings were provided with ball joints to simulate the typical pinned boundary conditions. The  
6 intended elevated load was achieved by means of lead ballast plates mounted on the model deck. The  
7 elevated mass was free to vibrate in all 6 degrees of freedom (DOF) under the leg excitation. Figure 7  
8 illustrates the jack-up model and the principal properties of the model are displayed in Table 1.



9  
10 Figure 7. Jack-up model

11

---

1 Table 1. Model properties

<b>Particulars</b>	<b>Prototype</b>	<b>Model</b>
Length (m)	20.00	0.71
Breadth (m)	16.00	0.57
Longitudinal Leg spacing (m)	16.00	0.57
Transverse Leg Spacing (m)	12.00	0.43
Elevated Load (kg)	400,000	17.94
Leg cantilever length (m)	27.35	0.97
Leg Diameter (m)	0.94	0.034
Leg Material	Mild Steel	PVC
Hull Material	Mild Steel	Plastic
Water Depth (m)	20.00	0.78
Wave Height (m)	2.00	0.07
Current (knots)	4.00	0.75

2

### 3 3.2. Experimental Setup

4 The experiment was conducted in the wind wave current (WWC) tank at the Hydrodynamic Laboratory  
5 of the School of Marine Science and Technology, Newcastle University. The WWC tank has a length of  
6 11.00 m, a width of 1.80 m and can accommodate a maximum water depth of 1.00 m, maximum wave  
7 height of 0.12 m and a maximum current of 1.00 m/s. The model was fixed to the tank bottom, positioned  
8 at the measuring section of the WWC tank and was exposed to currents with various speeds. The model  
9 was mounted on a turntable base which can be rotated to simulate various headings. The model responses  
10 were measured by means of 2 Qualisys motion tracking cameras installed on either side of the model. The  
11 experimental setup is illustrated in Figure 8.

12 The tank blockage is 3.78%, and the distance of the model leg from the nearest side wall is around 20  
13 times its diameter. This ensures insignificant flow blockage and boundary effects on the test results  
14 (Chakrabarti, 2005). The scale effect due to the difference in model and prototype Reynolds numbers is  
15 expected to be minimal due to the free vibrations of the legs, as even small vibrations tend to synchronise



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1 vortex shedding and separation points (DNV.GL, 2017). Moreover, the Reynold's number influence  
2 experienced by a smooth stationary cylinder in an incoming laminar flow is not significantly felt in real  
3 flows with substantial incoming turbulence, as verified with full scale submarine pipelines and piles  
4 experiencing aggressive cross flow VIV even in critical and supercritical regimes (Raven et al., 1985;  
5 Sainsbury and King, 1971). The full scale conditions around a jack-up leg are usually influenced by a  
6 reasonable surface roughness due to the presence of marine growth (PANEL OC-7, 2008). The high  
7 turbulence intensity of around 13% generated by the inlet grid ensures a turbulent boundary layer during  
8 test conditions in the subcritical flow regime, similar to the realistic prototype conditions (Chakrabarti,  
9 2005).



10

11 Figure 8. Experimental setup

---

### 1 3.3. Test Procedure

2 The tests involved mass tests, stiffness tests, free decay test and response tests. Mass tests yielded the  
3 mass, inertia and centre of gravity (COG) details of the model. Stiffness tests verified the stiffness of the  
4 model, the effect of ball joint friction on the model stiffness and associated nonlinearities. Natural  
5 frequencies, damping properties and the dynamic nonlinearities of the models were established with the  
6 free decays tests. Response tests were carried out to evaluate the modes, amplitudes and frequencies of  
7 the VIV responses experienced by the model. It is noted that response tests were carried out both in  
8 uniform current and wind to verify the effect of mass ratio and to validate the criteria developed. In order  
9 to capture all the operational envelope of a jack-up, response tests were conducted for three water depths  
10 and three elevated loads. The effect of the VIV on the mean drag force acting on the model was also  
11 established with the response tests. The details of the test cases are presented in Table 2.

12 Table 2. Test Matrix

---

<b>Test Case</b>	<b>Description</b>	<b>Hull Mass (kg)</b>	<b>Fluid Medium</b>	<b>Water Depth (m)</b>	<b>Effective Water Depth (m)</b>
NVL, Dry	No Variable Load	10.65	Air	-	-
VL, Dry	Full Variable Load	18.05	Air	-	-
VL, 500WD	Full Variable Load	18.05	Water	0.50	0.39
VL, 700WD	Full Variable Load	18.05	Water	0.70	0.59
NVL, 890WD	No Variable Load	10.65	Water	0.89	0.78
RVL, 890WD	Reduced Variable Load	14.08	Water	0.89	0.78
VL, 890WD	Full Variable Load	18.05	Water	0.89	0.78

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13

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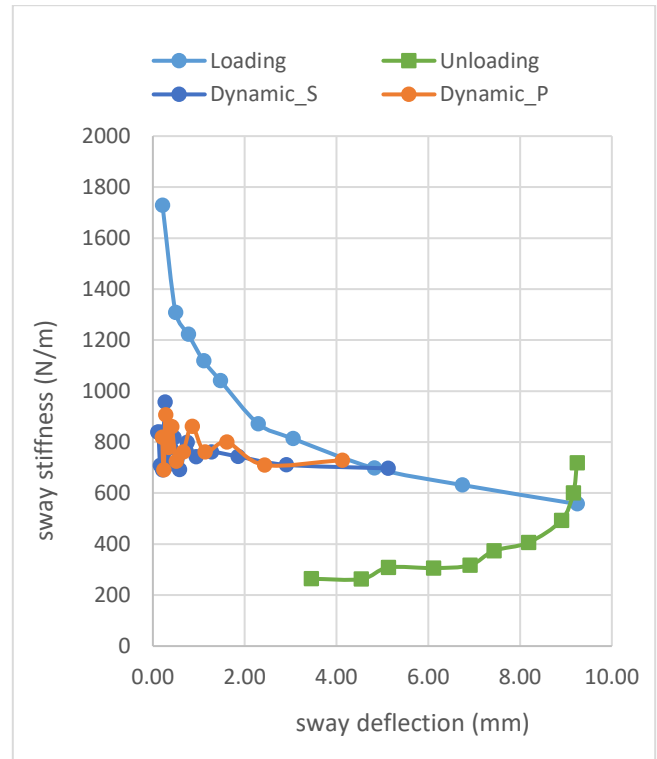
## 1 4. Results and Discussion

### 2 4.1. Stiffness Tests

3 Figure 9a) illustrates the sway stiffness test setup and 9b) demonstrates the variation of sway stiffness  
4 with the deflection for the test case NVL, 890WD. It is observed that the ball joint friction was influencing  
5 the rotational fixity of the leg footing. Leg bottom behaved as a fixed footing for very small initial  
6 deflections and for reasonably large deflections as pinned footing. The model stiffness was also found to  
7 reduce during unloading cycle due to the reversal of ball joint friction. The mentioned behaviours are  
8 illustrated in the stiffness variations in Figure 9b), which also demonstrates the corresponding dynamic  
9 sway stiffnesses for the port (P) and starboard (S) sides, derived from the free decay tests. It can be  
10 observed that the dynamic stiffness is less dependent on the deflection and nearly corresponds to that of  
11 pinned footing. The surge and sway stiffnesses were found to be identical as expected due to the symmetric  
12 configuration. Yaw stiffness tests also exhibited footing fixity at very small deflections similar to the surge  
13 and sway results.

### 14 4.2. Free Decay Tests

15 Free decay tests were carried out with the model in air as well as in still water. The tests were intended to  
16 establish the natural frequency, added mass, damping and the nonlinearities of the system. As the model  
17 was found to exhibit softening nonlinearity at large deflections due to the ball joint friction or support  
18 rotational stiffness, the free decay time series was truncated to remove large amplitudes, and Fast Fourier  
19 Transform (FFT) analysis was carried out to establish the natural frequencies. The added mass of the legs  
20 was calculated from the differences in the natural frequencies of the model in air and water. The variation  
21 of the damping factor with the natural frequencies was calculated from the successive amplitude decays,  
22 and the trend was used to calculate the damping ratio corresponding to the initial natural frequency.



a)

b)

Figure 9. a) Sway stiffness test setup b) Sway stiffness variation with deflection for NVL, 890WD

1 Figure 10 illustrates the results of the sway free decay test of the model for the test case, NVL, 890WD.

2 Figure 10)a, 10)b and 10)c displays the result of FFT of the truncated histogram, the variation of the

3 natural frequency with sway deflections and the variation of the damping ratio with the natural frequency.

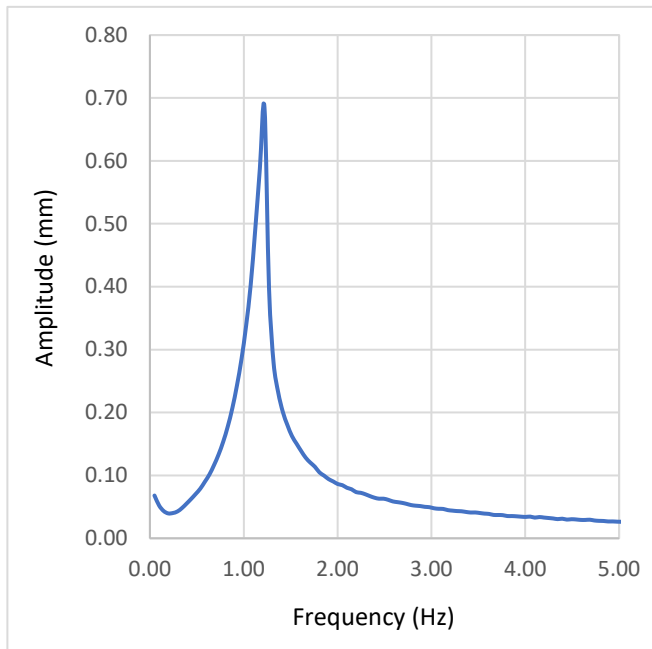
4 It was found that the damping of the model increases with reducing the natural frequency or increasing

5 vibration amplitudes. However, it is worth noting that the damping ratios obtained from the tests were

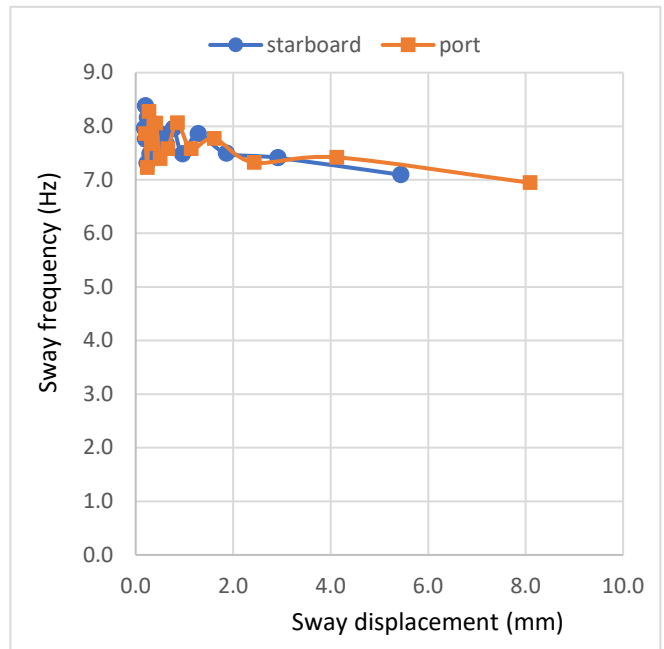
6 found to exhibit considerable scatter owing to the ball joint friction which influenced the results as dry

7 friction or coulomb damping.

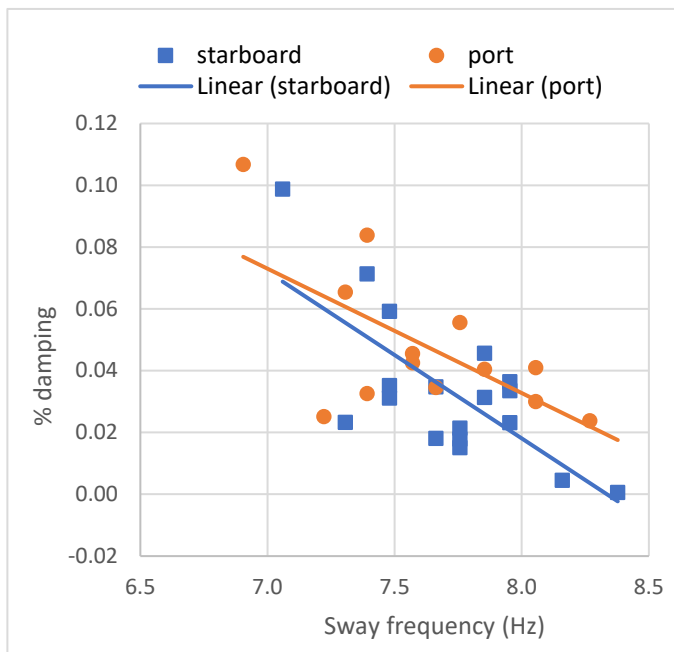
8



a)



b)



c)

Figure 10. a) FFT of sway free decay test; NVL, 890WD b) Sway natural frequency variation with displacement; NVL, 890WD c) Variation of sway damping ratio with sway frequency; NVL, 890WD.

- 1 Tables 3 and 4 summarise the results of sway and yaw free decay tests respectively. From free decay tests,
- 2 the added mass coefficient of the legs in still water was found to be around 1, approaching the theoretical

1 value. The corresponding mode shape was found to be that of pinned footing. This showed that for the  
 2 practical vibration amplitudes, the model legs were vibrating with a mode shape near to that of pinned  
 3 footing and the effect of the ball joint friction was negligible. It was also revealed that the damping of the  
 4 system had increased in still water due to the additional fluid damping.

5 Table 3. Model natural frequency, added mass and damping from sway free decay tests

Sway							
Test Case	Damping Ratio	Natural Frequency (Hz)	Natural Period (s)	Added Mass (kg)	Leg Added Mass Coefficient	Fluid Damping (Ns/m)	Fluid damping ratio
NVL, Dry	0.036	1.37	0.73	0.00	-	0.00	0.00
VL, Dry	0.031	1.03	0.97	0.00	-	0.00	0.00
VL, 500WD	0.041	1.00	1.00	0.12	0.71	2.09	0.01
VL, 700WD	0.035	0.98	1.02	0.56	1.06	1.00	0.00
NVL, 890WD	0.040	1.22	0.82	1.46	1.35	1.15	0.01
RVL, 890WD	0.044	1.08	0.93	1.40	1.29	2.87	0.01
VL, 890WD	0.038	0.95	1.05	1.19	1.10	1.97	0.01

6 Table 4. Model natural frequency, added mass and damping from yaw free decay tests

Yaw							
Test Case	Damping Ratio	Natural Frequency (Hz)	Natural Period (s)	Added Mass Inertia (kgm <sup>2</sup> )	Leg Added Mass Coefficient	Fluid Damping (Nms/rad)	Fluid damping ratio
NVL, Dry	0.037	2.05	0.49	0.00	-	0.00	0.00
VL, Dry	0.040	1.91	0.52	0.00	-	0.00	0.00
VL, 500WD	0.041	1.86	0.54	0.02	1.00	0.04	0.00
VL, 700WD	0.042	1.78	0.56	0.07	1.00	0.10	0.00
NVL, 890WD	0.038	1.81	0.55	0.13	0.88	0.11	0.00
RVL, 890WD	0.043	1.74	0.58	0.14	0.95	0.21	0.01
VL, 890WD	0.043	1.72	0.58	0.12	0.82	0.15	0.01

7 Table 5 displays the mass ratio, mode factor, effective mass damping parameter and effective inertia  
 8 damping parameter of the model for the respective test cases. It can be inferred from the VIV criteria in  
 9 Equations (35), (38), (45) and (49) that model will not experience any of the VIV modes in wind, owing  
 10 to high effective mass and inertial damping parameters in the air. Further, the model is expected to  
 11 experience cross flow VIV and yaw VIV due to cross flow excitation during all the test cases in water. By

---

1 comparing the effective mass damping and inertia damping parameters, it is expected that the responses  
2 increase with increasing water depth and will be highest for the lightest test case, ‘NVL, 890WD’.

3 On verifying against the VIV criteria in Equations (35), it can be noted that the model is expected to  
4 experience inline VIV only for the lightest two test cases, ‘RVL, 890WD’ and ‘NVL, 890WD’ at 0.89m  
5 water depth. Similarly, verification against Equation (45) reveals that the model is expected to experience  
6 yaw VIV due to inline excitation for all the three test cases at 890m water depth. However, it can be  
7 deciphered that the  $Re$  corresponding to the inline VIV and yaw VIV due to inline excitation for the test  
8 cases in water is around 3000 to 5000, and the corresponding stationary cylinder oscillatory drag  
9 coefficient is 0.06 (Bartrop and Adams, 1991) that is considerably less than the upper limit value of 0.10  
10 considered in Equation (35) and (45). On proportionally correcting the right hand side (RHS) of the  
11 equations and verifying against Table 5, it can be observed that the model will not be vulnerable to inline  
12 VIV and yaw VIV due to inline excitation in any of the test cases.

### 13 4.3. Response Tests

14 Response tests were carried out by exposing the model to incremental current from the zero heading (0  
15 degree angle of attack). Three drafts and three loading conditions were considered for the tests to cover  
16 the realistic operation scenarios and the practical ranges of the mass ratio and the mass damping parameter.  
17 The responses of the hull along the 6 DOF were measured and recorded as individual time series. The  
18 inline, transverse and yaw time series were post processed, and the root-mean-square (rms) amplitudes  
19 and vibration frequencies were calculated for various current speeds. The results of the response tests are  
20 also summarised in Table 5.

1 Table 5. Response test results, a) cross flow/sway b) yaw

a)

Sway								
Case	$m^*$	MF	$\zeta$	$m^*\zeta$	MF	$V_{ry}$	$y_{O,rms}/D$	$y_{O,max}/D$
VL, Dry	6719.10	1.571	0.031	328		4.80	0.00	0.00
VL, 500WD	13.65	3.274	0.041	1.82		4.80	0.02	0.03
VL, 700WD	9.38	2.276	0.035	0.75		4.92	0.04	0.06
VL, 890WD	7.38	1.812	0.038	0.51		5.05	0.06	0.10
RVL, 890WD	5.96	1.812	0.044	0.47		4.82	0.07	0.11
NVL, 890WD	4.73	1.812	0.040	0.34		4.84	0.08	0.13

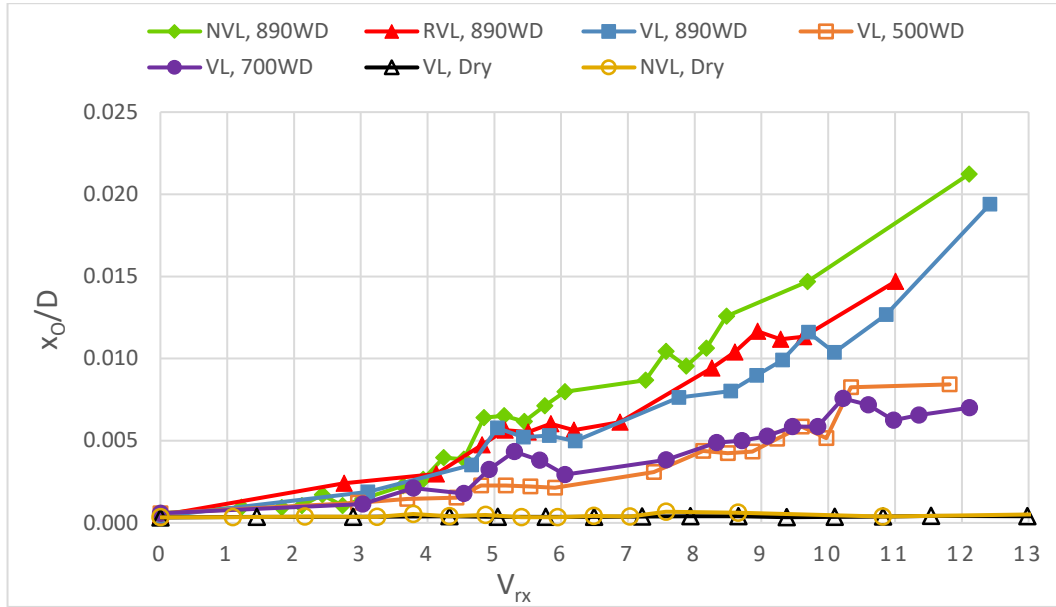
b)

Yaw								
Case	$m^*$	MF	$\zeta$	$r_\phi$ (m)	$m^*\zeta$ MF $r_\phi^2$ (m <sup>2</sup> )	$V_{r\phi}$	$R_{\phi\phi_{O,rms}}/D$	$R_{\phi\phi_{O,max}}/D$
VL, Dry	6719.10	1.571	0.040	0.215	19.2911	4.96	0.00	0.00
VL, 500WD	13.65	3.274	0.041	0.218	0.0863	4.96	0.02	0.04
VL, 700WD	9.38	2.276	0.042	0.224	0.0447	5.40	0.04	0.08
VL, 890WD	7.38	1.812	0.043	0.233	0.0310	5.15	0.07	0.13
RVL, 890WD	5.96	1.812	0.043	0.252	0.0293	5.54	0.06	0.12
NVL, 890WD	4.73	1.812	0.038	0.275	0.0245	5.32	0.06	0.11

2 4.3.1. *Inline and Cross flow VIV*

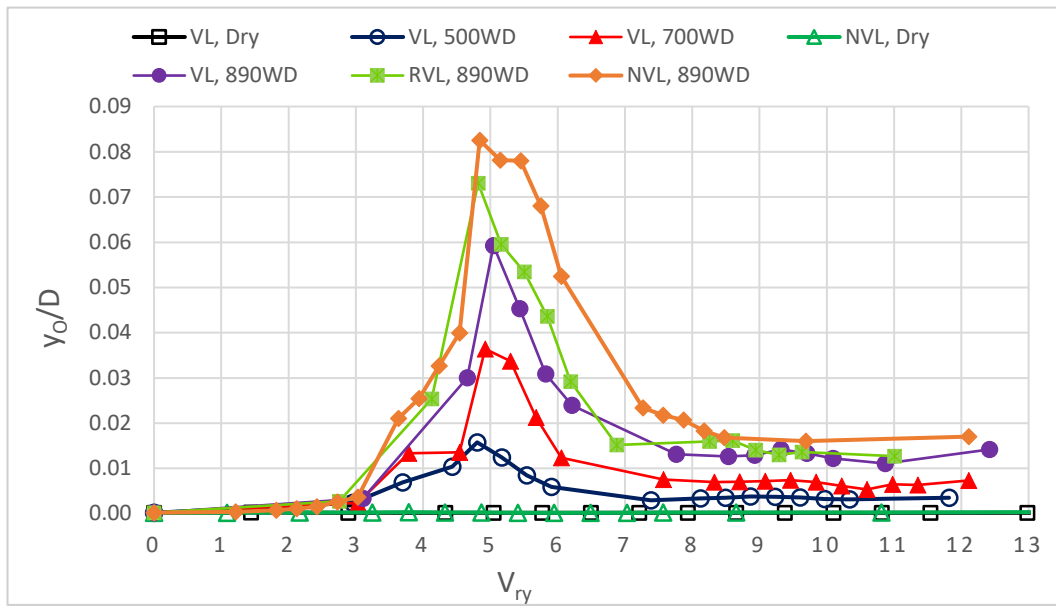
3 Figure 11 and 12 show the variation of the normalised inline and cross flow rms amplitude responses  
4 respectively, plotted with respect to the corresponding reduced velocities. It is observed that the model  
5 does not exhibit inline VIV response for any of the test cases as predicted and evident in Figure 11. This  
6 validated the mathematical model, inline VIV criteria, and the effect of high inherent mass ratios of the  
7 jack-ups. These results demonstrated the significance of the effective mass damping parameter in  
8 suppressing VIV.





1

2 Figure 11. Inline amplitude response with corresponding reduced velocity ( $U/f_{Nx}D$ )



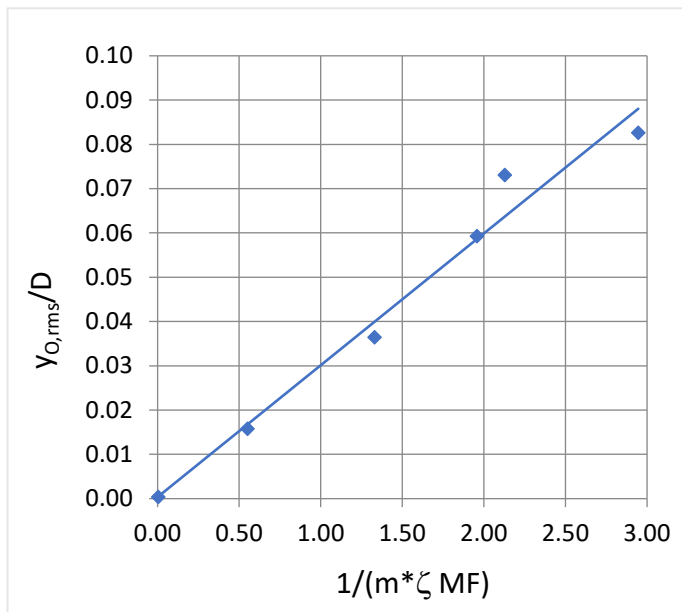
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4 Figure 12. Cross flow amplitude response with corresponding reduced velocity ( $U/f_{Ny}D$ )

5 Figure 12 evidently shows that the model was highly vulnerable to cross flow VIV with large lock-in  
 6 regimes. The response amplitudes (rms) were found to be around 0.08D, and the maximum response  
 7 amplitude was observed to be as high as 0.13D. The sway lock-in regime was found to extend between

1 reduced velocities of 3 to 8. Further, the response amplitude and lock-in range were found to increase with  
2 the increase in water depth and reduction in elevated load, with the lightest test case exhibiting the largest  
3 cross flow response amplitude and lock-in regime.

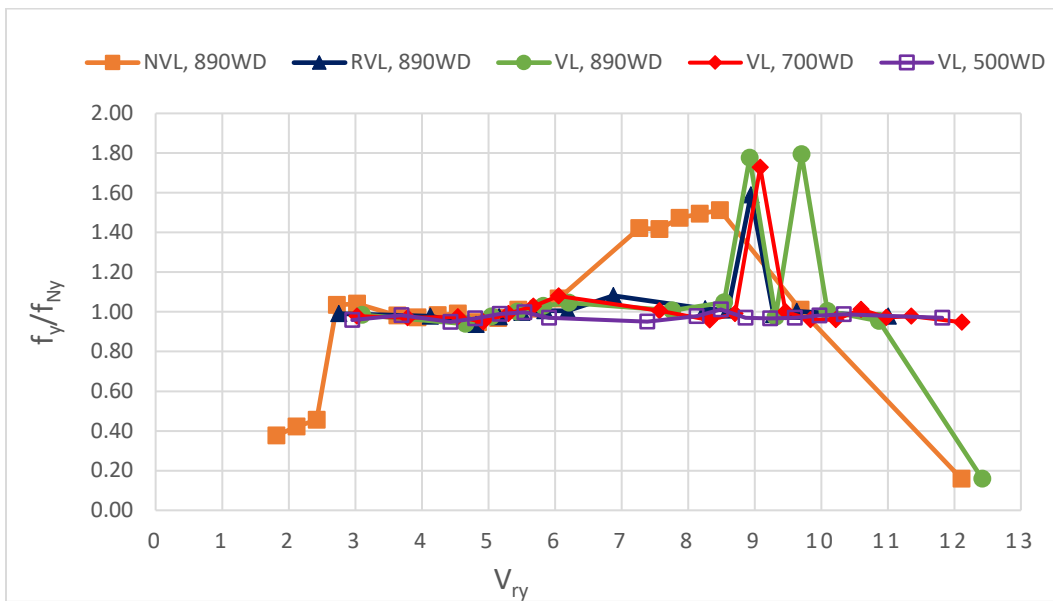
4 However, it is noted that the model does not exhibit any cross flow VIV during the response tests in wind.  
5 This validated the cross flow VIV criterion, the mathematical model and revealed the significance of the  
6 mass ratios in containing the VIV of the jack-ups. It was observed that the increase in mass ratio reduces  
7 the amplitude response and the lock-in range. The normalised cross flow rms response amplitude ratios  
8 fitted reasonably well in a straight line when plotted against the inverse of the effective mass damping  
9 parameter as shown in Figure 13. This further demonstrated the significance of the effective mass damping  
10 parameter in controlling the cross flow VIV response. Thus, the effective or modified mass damping  
11 parameter can be considered as the universal parameter for the comparison of cross flow VIV across  
12 various mode shapes and water depths.



13

14 Figure 13. Cross flow amplitude response with the inverse of modified mass damping parameter.

1 The frequency responses of the models were calculated from the time series, normalised with the  
 2 respective natural frequencies and plotted against the corresponding reduced velocities. Figure 14  
 3 illustrates the cross flow frequency responses of the model for all the test cases. It can be clearly observed  
 4 that there exists two separated lock-in regimes, i.e., a lower regime with a frequency ratio of around unity  
 5 and an upper regime with a frequency ratio of above 1.40. In comparison with Figure 12, it is clear that  
 6 the lower regime corresponds to the cross flow lock-in responses in the sway mode while the upper regime  
 7 corresponded to lock-in responses in the yaw or torsional mode. It is noted that the cross flow vibrations  
 8 in the upper regime are coupled vibrations with the yaw lock-in frequencies. The coupling is evidently  
 9 strong and continuous for the lightest test case but weak and intermittent for the other cases.

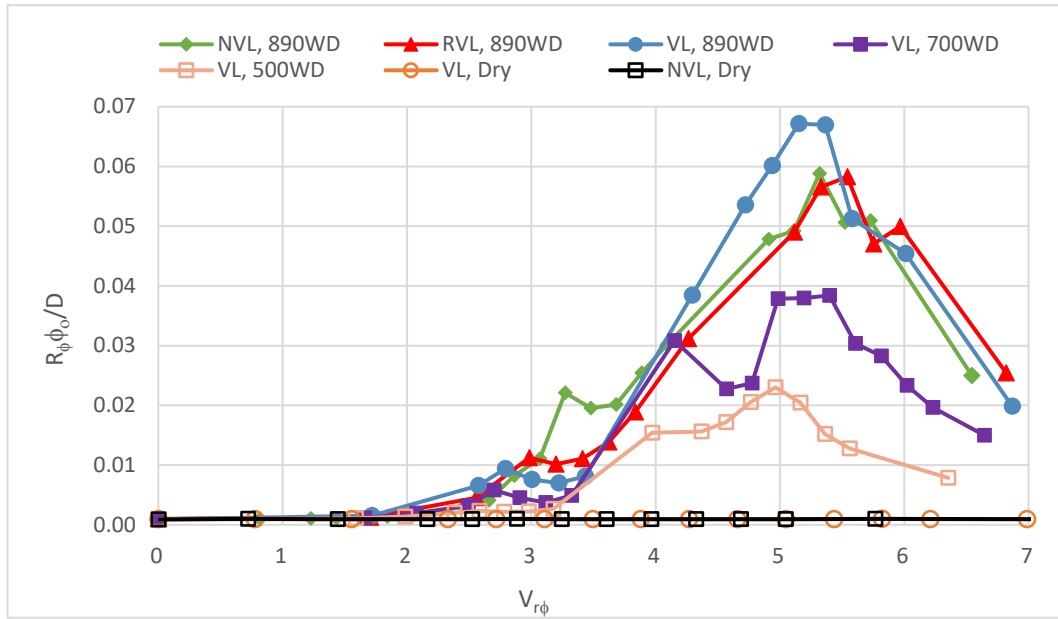


11 Figure 14. Cross flow frequency response with corresponding reduced velocity ( $U/f_{Ny}D$ )

12 4.3.2. Yaw VIV

13 Experimental observation revealed that the model jack-up experienced torsional or yaw VIV response  
 14 about the vertical axis at higher flow speeds. The yaw amplitude response as a function of yaw reduced  
 15 velocity is plotted in Figure 15. It can be seen that the peak yaw response occurs around a yaw reduced

1 velocity of 5, which demonstrates that the yaw VIV is due to cross flow lift excitation. The rms response  
 2 amplitudes are found to be around  $0.07D$ , and the maximum response amplitude is around  $0.13D$ . The  
 3 lock-in regime is found to extend between the yaw reduced velocities of 3 to 7. Yaw VIV due to inline  
 4 excitation was not observed in any of the test cases as anticipated, validating the respective yaw VIV  
 5 criterion and the mathematical model.

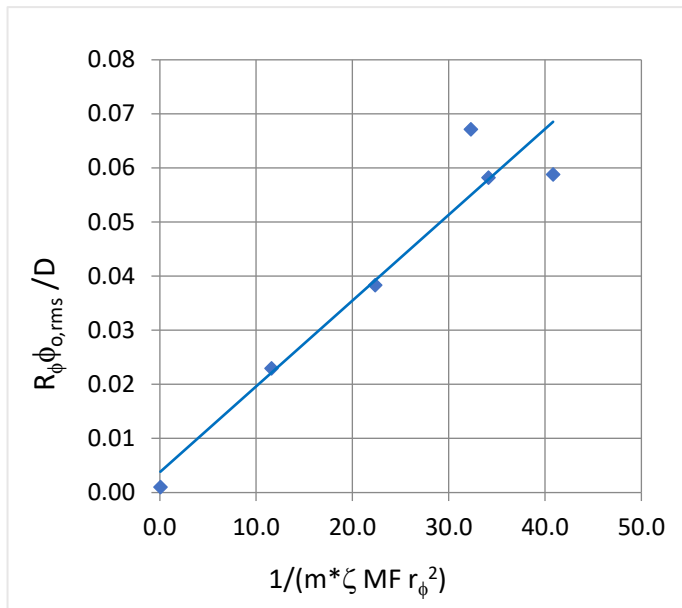


6  
 7 Figure 15. Yaw amplitude response with corresponding reduced velocity ( $U/f_N\phi D$ )

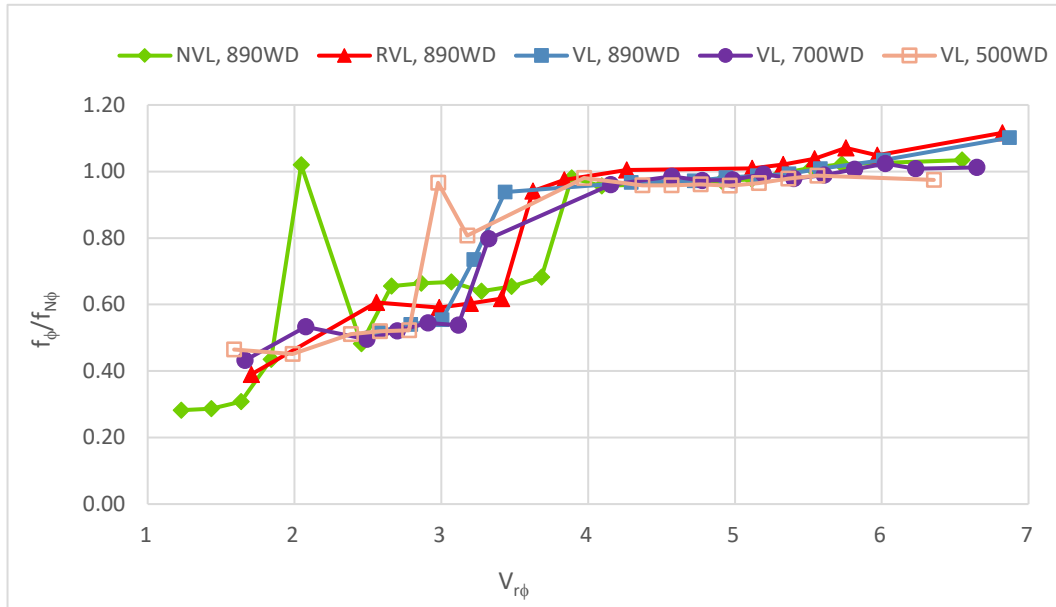
8 The response amplitude and lock-in range are found to increase generally with an increase in water depth.  
 9 However, amplitude and lock-in range do not increase with the reduction in elevated load, with the  
 10 response amplitudes displaying saturation in the lighter test cases. This behaviour may be attributed to the  
 11 increase in yaw radius of gyration of the model with decreasing elevated load, increase in the range of  
 12 overlapping sway vibrations and enhanced frequency coupling of sway with yaw vibrations with the  
 13 reduction in mass ratio. Hence further experiments are necessary with model experiencing pure yaw  
 14 vibrations and restrained along the cross flow direction.

1 It is noted that the model did not exhibit any yaw VIV either, as anticipated during the response tests in  
 2 wind demonstrating the validity of the mathematical model and the significance of the mass ratios in  
 3 containing the VIV of the jack-ups. It was observed that the increase in mass ratio reduces the amplitude  
 4 response and the lock-in range except for the test cases at maximum water depth. The response amplitude  
 5 ratio fitted reasonably well in a straight line except for the lighter test cases, when plotted against the  
 6 inverse of the effective inertia damping parameter as illustrated in Figure 16. Effective inertia damping  
 7 parameter is the product of mass damping parameter, mode factor and square of yaw radius of gyration,  
 8 which can be considered as the universal parameter determining the yaw VIV response.

9 The yaw frequency responses of the model normalised with the corresponding natural frequencies and  
 10 plotted against the yaw reduced velocities are displayed in Figure 17. Similar to cross flow frequency  
 11 response, the yaw frequency response of the model also revealed two separated lock-in regimes, a lower  
 12 regime with a frequency ratio below 0.70 and an upper regime with a frequency ratio of around 1. In  
 13 comparison with Figure 14 and Figure 15, it can be seen that the lower regime corresponds to the coupled  
 14 yaw vibrations in cross flow sway lock-in regime and the upper regime represents lock-in responses in the  
 15 yaw mode.



1 Figure 16. Yaw amplitude response with the inverse of effective inertia damping parameter

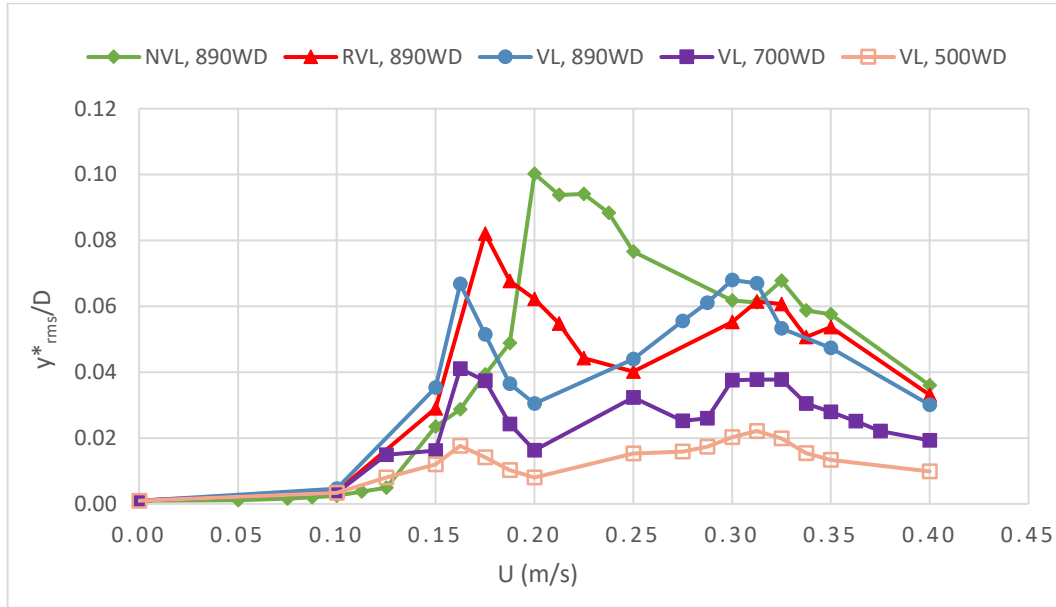


2

3 Figure 17. Yaw frequency response with corresponding reduced velocity ( $U/f_N\phi D$ )

#### 4 4.3.3. Combined Response

5 The combined cross flow amplitude response of the leg derived from the corresponding cross flow and  
6 yaw responses is plotted in Figure 18. It is found that the range of the yaw response overlaps with that of  
7 the cross flow response particularly for low mass ratios, causing a combined lock-in range throughout the  
8 operating current speeds. The combined vibrations experienced higher response amplitudes than the  
9 individual values, especially for low mass ratios. Further the combined lock-in range is seen extending  
10 almost throughout the operating current range (0.10 m/s to 0.40 m/s) making the jack-up practically  
11 redundant.

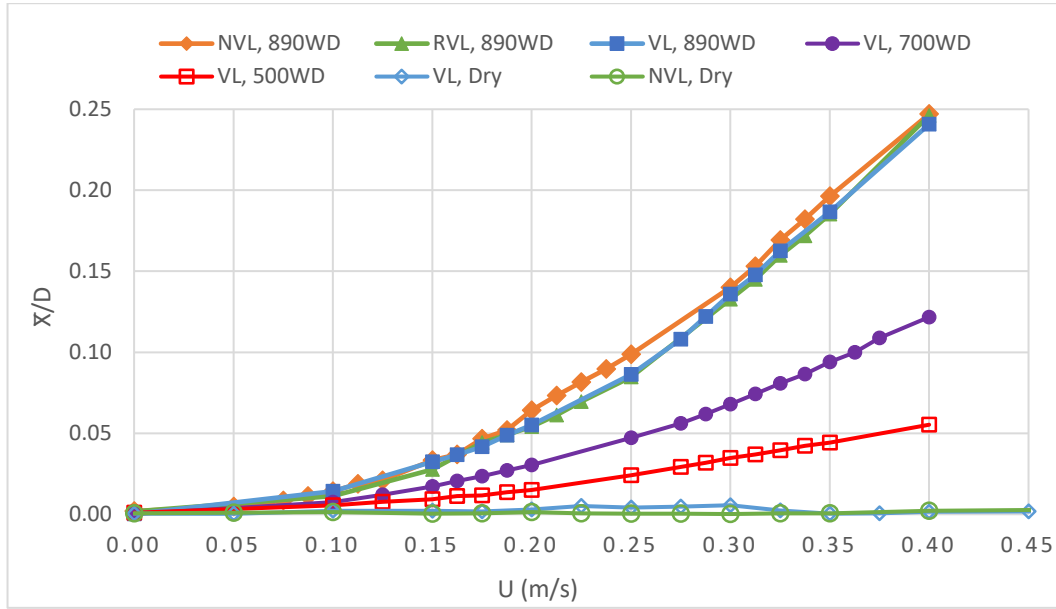


1

2 Figure 18. Leg cross flow combined amplitude response due to sway and yaw with current

3 *4.3.4. Mean Response*

4 The mean or steady inline responses were also measured as a function of current speeds, and the  
 5 normalised results were plotted in Figure 19. It can be observed that the mean inline response  
 6 approximately follows a quadratic variation with the flow speeds and hence can be considered as a good  
 7 representation of the drag force acting on the jack-up. The lightest operating condition (NVL, 890WD) at  
 8 the maximum water depth exhibited the highest mean inline response despite having a greater stiffness  
 9 due to lesser P delta effect. It is evident that the jack-up VIV increases the mean inline responses at all  
 10 practical speed ranges, clearly indicating an amplification of the mean drag acting on the legs. On the  
 11 other hand, the mean cross flow and yaw responses were found to be insignificant throughout the tested  
 12 current speeds.



1

Figure 19. Inline mean response of the model with current

2

#### 4.4. Recommendations

3

Based on the test results, it is recommended that the mass ratio of the jack-ups shall be as high as possible to minimise any potential VIV occurrence. Since the combined lock-in range of both the modes extends over almost all the practical current speeds, VIV and mean drag amplification can have a considerable effect on the yield and fatigue strength of the structure. Hence, necessary modifications to the present classification rules or guidelines are warranted to adequately account for the effect of VIV in the design of jack-ups.

9

#### 5. Conclusions

10

A simple mathematical model was developed based on the single-degree-of-freedom analogy and the principle of conservation of energy, which can be used to evaluate various modes of VIV of a jack-up with cylindrical legs in steady flow. Mass ratio, damping ratio and mode factor were found to be the

13



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1 important parameters controlling the inline and cross flow VIV of jack-ups. The radius of gyration was  
2 found to be an additional important parameter influencing yaw VIV. Criteria for the occurrence of inline,  
3 cross flow and yaw VIV were developed for the cases of a single 2D cylinder, four 2D cylinders in  
4 rectangular configuration and a complete jack-up with cylindrical legs in steady uniform flow.

5 The model tests demonstrated that the jack-up experienced both cross flow and yaw lock-in vibrations due  
6 to lift excitation, with maximum amplitude ratios in excess of 0.1D. The jack-up was found not to  
7 experience inline and yaw VIV due to oscillatory drag excitation. Experiments conducted in wind revealed  
8 that the jack-up was not experiencing any of the VIV modes and lock-in vibrations. The observed  
9 behaviours validated the newly developed VIV model, criteria and the importance of various parameters.

10 From the test results, it can be inferred that the jack-ups were highly vulnerable to cross flow VIV with  
11 large lock-in ranges in light operating conditions at high water depths. Further, the lock-in range of the  
12 cross flow VIV was found to overlap with the lock-in range of yaw VIV, particularly in light operating  
13 conditions making the unit almost redundant throughout the operating current ranges. The cross flow and  
14 yaw VIV were also observed to couple at higher current speeds causing very high combined leg amplitude  
15 response. From the mean inline displacements, it was also revealed that both the cross flow and yaw VIV  
16 increases the mean drag force acting on the jack-up.

17 The mathematical approach presented will enable practising engineers to effectively consider the effect  
18 of VIV in jack-up designs. The test results also underline the significance of yaw or torsional VIV in the  
19 case of rigidly coupled multi-cylinder structures.

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5 Cybermarine Technologies.

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## 1 7. Nomenclature

2	$a$	=	longitudinal or inline spacing of cylinders/legs
3	$b$	=	transverse or cross flow spacing of cylinders/legs
4	$C_L$	=	lift coefficient
5	$C$	=	damping coefficient
6	$C_d$	=	oscillatory drag coefficient
7	$C_\phi$	=	yaw damping coefficient
8	$C_{eL}$	=	effective SDOF damping coefficient per leg
9	$C_{e\phi}$	=	effective SDOF yaw damping coefficient
10	$D$	=	diameter of the cylinder
11	$d'$	=	effective water depth considering penetration
12	$f_{Nx}$	=	surge natural frequency
13	$f_{Ny}$	=	sway natural frequency
14	$f_{N\phi}$	=	yaw natural frequency
15	$f_v$	=	vortex shedding frequency of the cylinder
16	$f_O(z)$	=	oscillatory excitation force distribution along leg
17	$f_{Ox}(z)$	=	oscillatory drag force distribution along leg
18	$f_{Oy}(z)$	=	oscillatory lift force distribution along leg
19	$F_d$	=	oscillatory drag force
20	$F_{eL}$	=	effective SDOF force excitation per leg
21	$I$	=	yaw inertia
22	$k_L$	=	structural wave number/mode shape of the leg
23	$K$	=	stiffness of a linear mass spring cylinder
24	$K_\phi$	=	yaw stiffness

---

1	$L$	=	length, effective leg length (from leg bottom to hull interface)
2	$M$	=	leg mass distribution
3	$m^*$	=	mass ratio (mass per displaced mass) of the cylinder.
4	$M$	=	mass of a linear mass springs cylinder
5	$M_{eL}$	=	effective SDOF mass per leg at the hull interface level
6	$M_{\phi 0}$	=	Amplitude of oscillatory yaw moment
7	$M_{e\phi L}$	=	effective SDOF yaw excitation per leg
8	$M_{e\phi}$	=	effective SDOF yaw excitation
9	$MF$	=	Mode Factor
10	$m_{\phi}(z)$	=	oscillatory yaw moment distribution along leg
11	$m(z)$	=	mass distribution along leg
12	$Re$	=	Reynolds Number
13	$r_{\phi}$	=	yaw radius of gyration
14	$R_{\phi}$	=	radial distance of the leg from yaw centre
15	$St$	=	Strouhal number
16	$U$	=	2D steady flow or uniform current velocity
17	$V_{rx}$	=	surge reduced velocity
18	$V_{ry}$	=	sway reduced velocity
19	$V_{r\phi}$	=	yaw reduced velocity
20	$x_o$	=	amplitude of inline VIV response
21	$x_o(z)$	=	inline VIV response amplitude variation along leg
22	$X_L$	=	inline VIV response of leg in way of the hull interface
23	$y_o$	=	amplitude of cross flow VIV response
24	$Y_L$	=	cross flow VIV response of leg in way of the hull interface

- 
- 1  $z$  = elevation w.r.t to the seabed, positive upwards; at the seabed
  - 2  $\phi_O$  = amplitude of yaw response
  - 3  $\phi_L$  = yaw VIV response in way of the hull interface
  - 4  $\rho$  = density of the fluid
  - 5  $\omega_N$  = natural angular frequency
  - 6  $\omega_V$  = vortex shedding angular frequency
  - 7  $\zeta$  = damping ratio

---

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